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Wave power absorption by an oscillating water column 1 (OWC) device of annular cross-section in a combined wind-2 wave energy system 3 4 Peiwen Cong^a, Bin Teng^a, Wei Bai^b, Dezhi Ning^a, and Yingyi Liu^c 5 6 7 ^a State Key Laboratory of Coastal and Offshore Engineering, Dalian University of Technology, Dalian 116024, 8 China ^b Department of Computing and Mathematics, Manchester Metropolitan University, Chester Street, Manchester 9 10 M1 5GD, UK 11 ^c Research Institute for Applied Mechanics, Kyushu University, Fukuoka 8168580, Japan 12 13 14 Abstract This paper deals with a new combined concept consisting of an oscillating water 15 16 column (OWC) device and an offshore wind turbine for the multi-purpose utilization 17 of offshore renewable energy resources. The wind turbine is supported by a monopile foundation, and the attached OWC is coaxial with the foundation. Within the chamber, 18 four vertical stiffening plates connect the exterior shell of the OWC and the monopile 19 20 foundation. Correspondingly, the whole chamber is divided into four equivalent fanshaped sub-chambers. A higher-order boundary element method is then adopted to 21 model the wave interaction with the combined system. Numerical models based on two 22 23 different approaches, namely 'Direct' and 'Indirect', are both developed in this study. 24 In addition, a self-adaptive Gauss integration method is developed to treat the nearly singular integration that occurs when the field and source points are very close to each 25 other. A detailed numerical analysis is then conducted for the case of an OWC 26 integrated into a NREL 5 MW wind turbine in both regular and irregular sea states. 27 28 Numerical results illustrate that a significant energy extraction efficiency is attained when remarkable piston-like fluid motion is induced within each sub-chamber, and the 29 30 wave power absorption by the OWC is not restricted by the wave direction. The air compressibility makes a negative effect on the wave power absorption especially when the wave frequency is less than the resonance frequency of the piston-mode motion of the fluid in the chamber. In addition, the wave forces on the OWC and the monopile can balance each other at specific wave conditions, leading to a nearly zero net wave force on the whole system. The results also illustrate that by using an optimal turbine parameter, the wave power production by the OWC can be an important supplement to the combined system in operational sea states.

38

Keywords: combined concept; wind turbine; monopile foundation; oscillating water
column (OWC); HOBEM

41

42 1. Introduction

43 The ocean is vast and powerful, enabling marine renewable energy to be potentially a significant energy supply. Due to the high power density and longtime availability, 44 considerable efforts and advances have been made in exploiting the power of ocean 45 waves worldwide, and a variety of wave energy converters have been invented to 46 harvest the wave energy. Among different classes of designs, the oscillating water 47 column (OWC) device has been widely regarded as one of the most promising options 48 49 [1]. A typical OWC device mainly consists of a collector chamber with an underwater bottom open to the sea and a power take-off (PTO) system, mostly an air turbine, on 50 51 the roof of the chamber [2]. In addition, in a typical OWC, the moving mechanical part is only the air turbine, located above the water surface. Due to the nature of simplicity, 52 the OWC is flexibly adapted to the shoreline, nearshore, and offshore through different 53 54 forms.

In the case of an onshore OWC, the wet power-transmission cables and mooring lines are not required. In addition, onshore OWCs can be easily maintained and normally operate in safe sea environments, which increases their survivability. Due to the apparent advantages, onshore OWCs have attracted much attention from researchers, and there have been many attempts on the hydrodynamic aspects of onshore OWCs [3–

7]. As the waves approach the shore, a considerable amount of wave energy can be 60 consumed due to the bottom friction. In addition, the shoreline location introduces 61 limitation to the deployment of large numbers of devices. Therefore, offshore OWCs, 62 which can be exposed to a higher amount of wave energy, have also been suggested and 63 designed by many researchers [8–10]. Offshore OWCs are in general floating devices, 64 65 with a typical geometry consisting of a truncated vertical column with an open bottom. Regarding random ocean waves, broadening the effective frequency bandwidth of an 66 67 OWC device is of great importance for its adaption to variable ocean environments. Therefore, researchers [11-17] have reported innovative configurations of the OWC 68 device, such as dual-chamber OWCs and OWC arrays. In the meantime, as the cost 69 sharing between the energy extraction and coastal protection can help improve the 70 economic viability, the idea of the integration of an OWC device into a breakwater has 71 also been proposed by researchers, such as [18–20]. 72

In addition to wave energy, wind energy is also a great source of renewable energy. 73 Wind energy converters have been used for the harvesting and exploitation of the 74 75 available enormous wind energy resources. Offshore wind turbine technology has been being developed rapidly in recent years, and it can be considered as the leading 76 technology in the offshore renewable energy sector. For both the offshore wind turbine 77 and the wave energy converter, there is a need for a reduction in the cost and the further 78 79 development. Due to the natural correlation, wave energy may also be of considerable amount where the offshore wind energy resource is rich. Significant opportunities and 80 81 benefits have been identified through an integration of the energy systems of different technologies into one single platform. The combination of the wind and wave energy 82 83 devices can reduce the cost relevant to the operation and maintenance, foundation substructure as well as the required electric grid infrastructure especially with regard to 84 the transportation of the produced power to onshore stations [21]. Except for the good 85 aspects from a cost-benefit point of view, the combination of the two energy systems 86 can also lead to the efficient use of the ocean space [22]. Due to the various possible 87 advantages as a result of the combination of the wind and wave energy converters, 88 several concepts of the combined system have been proposed, and the concept 89

feasibility of different integrated platforms has been assessed. Examples include the
integration of a point-absorber-type wave energy converter with a semisubmersible type
or a spar type wind turbine [21, 23-25]; integration of an OWC with a floating or
bottom-mounted offshore wind turbine [26-31], and semisubmersible flap concept [22,
32-33].

95 So far, the most widely used support structure for an offshore wind turbine is the monopile foundation, accounting for 87% of the installed wind farms in Europe [34]. 96 97 In the meantime, among various wave energy converters, the OWC device has been considered as one of the most promising options, reaching the stage of full-scale 98 prototypes. It suggests that the integration of an OWC with a monopile supporting a 99 wind turbine can be a promising solution for the multi-purpose utilization of offshore 100 renewable energy. For structures of fundamental geometry, analytical solutions can be 101 102 achieved. An analytical solution has been developed in [29] for a combined wind-wave energy system, in which the exterior shell of the attached OWC has a skirt whose scope 103 is to guide the wave energy flux inside the chamber. In the meantime, for structures of 104 105 complex geometry, it is quite difficult to achieve analytical solutions. Then, a numerical approach has to be developed. In this study, a novel combined concept consisting of an 106 oscillating water column (OWC) device and an offshore wind turbine is proposed. As 107 shown Fig. 1, the wind turbine is supported by a monopile foundation, and the OWC is 108 109 coaxial with the foundation. The OWC is partly submerged with its bottom open to the sea. An air duct, which houses an air turbine, is installed on the roof of the chamber. 110 Within the chamber, the exterior shell of the OWC and the monopile foundation are 111 connected by four vertical stiffening plates. Correspondingly, the whole chamber is 112 divided into four fan-shaped sub-chambers. The performance of the proposed system 113 is then investigated. In order to achieve this, numerical models based on HOBEM are 114 developed to analyze the wave interaction with the combined system. Based on the 115 developed models, a detailed numerical study is then conducted, and the wave power 116 117 absorption by the combined system in both regular and irregular sea states are analyzed. The remaining part of the paper is organized as follows. First, the hydrodynamic 118 problem and the power take-off model is introduced. In the next section, a solution to 119

the boundary value problem is developed, which is followed by the calculation of the volume flux, air pressure, optimal pneumatic damping coefficient, and wave power absorption. Then, after examining the convergence and validity of the proposed model, a detailed numerical analysis is conducted. Finally, conclusions are drawn based on the previous analysis.



125

126 Fig. 1 Overview of a combined concept consisting of a monopile wind turbine and an attached

127





128

Fig. 2 Definition of the coordinate system

131 **2.** Description of the hydrodynamic problem and the power take-off model

The hydrodynamic performance of a combined concept consisting of an OWC device 132 and an offshore wind turbine is considered. As shown in Fig. 1, the monopile is of radius 133 a, and the OWC is made coaxial with the monopile. Then, the monopile plays the role 134 of the interior shell of the OWC. The draft of the OWC is d. For the exterior shell of 135 the OWC, its inner and outer radii are R_i and R_e , respectively, with $e = R_e - R_i$ being the 136 thickness of the exterior shell. The chamber is connected with the external atmosphere 137 by a duct installed on the roof of the chamber, and there exists an air turbine housed in 138 the duct. Within the chamber, the exterior shell of the OWC and the monopile 139 foundation are connected by four vertical stiffening plates. The plates are with the 140 thickness e and draft d. The whole chamber of the OWC is then divided into four fan-141 shaped sub-chambers. At initial time, the free surfaces inside and outside the chamber 142 are at the same level, and there is an amount of air entrapped above the water surface 143 inside the chamber. The definition of the Cartesian and cylindrical coordinate systems 144 is given in Fig. 2. The oxy and $or\theta$ planes are both located on the mean plane of the free 145 surface, and the z-axis is oriented vertically upward. $r = \sqrt{x^2 + y^2}$ and $\theta = \tan^{-1}(y/x)$. 146 Two plates are installed along the x-axis, and the other two are along the y-axis. The 147 water depth *h* is a constant in this study. 148

It is assumed that the fluid is inviscid and incompressible with a constant density ρ , the fluid motion is irrotational, and the wave steepness is small. The linear potential flow theory can then be used, and there exists a velocity potential $\Phi(x; t)$ satisfying the Laplace's equation in the fluid domain. That is

153

$$\nabla^2 \Phi(\mathbf{x}; t) = 0. \tag{1}$$

The combined system is exposed to the action of a plane incident wave of amplitude A, and frequency ω . The wave heading is β with respect to the positive *x*-direction. A frequency-domain analysis is conducted, and all time-dependent variables are assumed to be harmonic. The time factor can then be separated, and the velocity potential will have the following form

$$\Phi(\mathbf{x}; t) = \operatorname{Re}\left[\phi^{(1)}(\mathbf{x})e^{-i\omega t}\right], \qquad (2)$$

160 in which 'Re' is the real part of a complex variable; $i = \sqrt{-1}$.

Besides Laplace's equation, the velocity potential must satisfy appropriate boundary 161 conditions. On the exterior free surface S_e ($r \ge R_e$), there is only the atmospheric 162 pressure P_0 . However, the inner free surface S_i ($a \le r \le R_i$) is subjected to an air pressure 163 distribution of $P_0 + P_c(t)$, in which $P_c(t)$ is the oscillating air pressure. Due to the high 164 sound speed in air and the low frequency of ocean waves, the oscillating air pressure 165 can be considered spatially uniform throughout the whole chamber [19]. Referring to 166 [35], after assuming isentropy and using a linear wave theory, the mass flux of the air 167 168 through a linear turbine is related to the oscillating air pressure by

169
$$\frac{\mathrm{d}M_a}{\mathrm{d}t} = \rho_a Q_c - \frac{V_0}{c^2} \frac{\mathrm{d}P_c}{\mathrm{d}t} = \frac{KD}{N} P_c, \qquad (3)$$

in which M_a is the mass of the air in the chamber; Q_c is the change rate of the total volume of air inside the chamber; V_0 is the air volume in the chamber in calm water; ρ_a is the air density; D is the diameter of the turbine rotor; N is the speed of the turbine rotation; c is the speed of the sound in air; K is an empirical coefficient depending on the design of the turbines [19]. For simple harmonic motions with

175
$$Q_c(t) = \operatorname{Re}\left[q_c e^{-i\omega t}\right]; \tag{4a}$$

176
$$P_c(t) = \operatorname{Re}\left[p_c e^{-i\omega t}\right], \tag{4b}$$

177 we can have

159

$$q_c = \Lambda p_c, \tag{5}$$

in which q_c and p_c are the amplitudes of the volume flux and the oscillating air pressure, respectively. The parameter Λ is expressed as

181 $\Lambda = \chi - i\omega\mu, \tag{6}$

182 with

183
$$\chi = \frac{KD}{N\rho_a};$$
 (7a)

184
$$\mu = \frac{V_0}{c^2 \rho_a}.$$
 (7b)

185 χ depends on the design of the air turbine, and can be adjusted by some ways, such as 186 varying the rotational speed *N*. μ represents the effect of compressibility of air in the 187 chamber, and is analogous to a spring constant [19]. The compressibility of air in a 188 chamber with a larger size can be more evident as discussed in Sheng et al. [36]. 189 Hereinafter, we define χ and μ as the turbine parameter and the chamber parameter, 190 respectively.

In the hydrodynamic analysis, the effect of the oscillating air pressure should be considered properly. Then, on the mean plane of the free surface (z = 0), the combined kinematic and dynamic boundary condition is given by

194
$$\frac{\partial \phi}{\partial z} - \frac{\omega^2}{g} \phi = \frac{i\omega}{\rho g} p_c, \text{ on } S_i;$$
(8a)

195
$$\frac{\partial \phi}{\partial z} - \frac{\omega^2}{g} \phi = 0, \quad \text{on } S_e.$$
(8b)

196 On the mean wet surface of the combined system S_b , the boundary condition is given 197 by

198
$$\frac{\partial \phi}{\partial n} = 0, \quad \text{on } S_b, \tag{9}$$

in which **n** is the normal unit vector pointing outward from the fluid domain. In the same way, the boundary condition on the impermeable sea bed (z = -h) is

201
$$\frac{\partial \phi}{\partial z} = 0, \quad \text{on } z = -h.$$
 (10)

To ensure the uniqueness of the solution, ϕ has to satisfy the Sommerfeld radiation condition at a substantial distance from the structure. That is

204
$$\lim_{r \to \infty} \sqrt{r} \left[\frac{\partial}{\partial r} (\phi - \phi_I) - i\kappa_0 (\phi - \phi_I) \right] = 0, \tag{11}$$

in which $\phi_I(\mathbf{x})$ represents the incident velocity potential, and it is given by

206
$$\phi_{I}(\mathbf{x}) = -\frac{iAg}{\omega} \frac{\cosh \kappa_{0}(z+h)}{\cosh \kappa_{0}h} e^{i\kappa_{0}(x\cos\beta+y\sin\beta)}, \qquad (12)$$

207 in which κ_0 is the wavenumber. κ_0 and ω satisfy the relationship $\omega^2 = g\kappa_0 \tanh \kappa_0 h$

208 , with g being the gravitational acceleration.

- 209 The captured power W_c is the time-averaged rate of work done by the oscillating air
- 210 pressure pushing the air through the air turbine [8]. W_c can be evaluated according to

211
$$W_{c} = \lim_{\hat{T} \to \infty} \left\{ \frac{1}{\hat{T}} \int_{t}^{t+\hat{T}} P_{c}(t) Q_{c}(t) dt \right\}.$$
 (13)

For regular incident wave, W_c can be further expressed as

213
$$W_c = \frac{1}{2} \operatorname{Re} \left[p_c q_c^* \right] = \frac{1}{2} \operatorname{Re} \left[\Lambda \right] \left| p_c \right|^2.$$
(14)

Following [37], the wave energy extraction efficiency E_c (also known as the relative capture width) for the present combined system is defined as

216
$$E_{c} = \frac{W_{c}}{W_{in}} = \frac{W_{c}}{2(R_{i} - a)P_{in}},$$
 (15)

in which W_{in} is the wave power of the free incident wave passing through the width of 2($R_i - a$) over a wave period; P_{in} , representing the power flux density of the incident wave, is determined according to

$$P_{in} = \frac{1}{2} \rho g A^2 C_g. \tag{16}$$

In Eq. (16), C_g is the group velocity of the incident wave.

222

220

3. Numerical approach to the boundary value problem

In order to achieve a solution to the boundary value problem defined in the previous sections, a decomposition of the volume flux is made. The volume flux inside the chamber is then expressed as a sum of two distinct parts. That is

$$q_c = q_D - \frac{i\omega}{\rho g} p_c q_R, \tag{17}$$

in which q_D and q_R represent the volume flux due to a wave diffraction problem and a pressure-dependent wave radiation problem, respectively. Correspondingly, the velocity potential is divided into two parts. That is

 $\phi = \hat{\phi} - \frac{i\omega}{\rho g} p_c \phi_R.$ (18)

232 In Eq. (18), $\hat{\phi}$ is a summation of the incident potential ϕ_i and the diffraction

233 potential ϕ_D , i.e.

234

$$\hat{\phi} = \phi_I + \phi_D; \tag{19}$$

235 ϕ_R is a pressure-dependent radiation potential.

The wave diffraction is caused by the excitation of the incident wave when the air 236 pressure inside and outside the chamber keeps the atmospheric pressure. The wave 237 radiation occurs when the wave motion is purely due to the oscillating air pressure 238 inside the chamber. The two problems can be regarded as the limited cases of the 239 problem discussed in the previous section without the effect of the oscillating air 240 pressure and the incident wave, respectively. ϕ_D and ϕ_R both satisfy the Laplace's 241 242 equation, a no-flow condition on the seabed, as well as a Sommerfeld condition in the far field. On the free surface, the following conditions are held by ϕ_D and ϕ_R . 243

244
$$\frac{\partial \phi_D}{\partial z} - \frac{\omega^2}{g} \phi_D = 0, \quad \text{on } S_i \text{ and } S_e, \tag{20}$$

245 and

246
$$\frac{\partial \phi_R}{\partial z} - \frac{\omega^2}{g} \phi_R = \begin{cases} -1, & \text{on } S_i; \\ 0, & \text{on } S_e. \end{cases}$$
(21)

247 On the body surface,
$$\phi_D$$
 and ϕ_R satisfy the following boundary conditions
248 $\frac{\partial \phi_D}{\partial n} = -\frac{\partial \phi_I}{\partial n}$, on S_b ; (22a)

249
$$\frac{\partial \phi_R}{\partial n} = 0, \quad \text{on } S_b.$$
(22b)

A boundary integral equation method is then used to solve the established boundary value problem, as it explicitly takes advantage of reducing the dimension of the problem by one order. The oscillating source, which satisfies a linear free-surface boundary condition, a no-flow condition on the horizontal seabed, and a Sommerfeld radiation condition at infinity, is used as Green's function. A mathematical expression for the Green's function $G(\mathbf{x}, \mathbf{x}_0; \omega)$ is given as follows

256
$$G(\mathbf{x}, \mathbf{x}_{0}; \omega) = -\frac{1}{4\pi} \left(\frac{1}{r_{1}} + \frac{1}{r_{2}} \right) + \frac{1}{4\pi} \int_{0}^{\infty} \frac{2(v+\mu)e^{-\mu h} \cosh \mu(z+h) \cosh \mu(z_{0}+h)}{v \cosh v h - \mu \sinh \mu h} J_{0}(\mu R) d\mu,$$

(23)

258 in which

259

$$R = \sqrt{\left(x - x_0\right)^2 + \left(y - y_0\right)^2};$$
 (24a)

260
$$r_1 = \sqrt{R^2 + (z - z_0)^2};$$
 (24b)

261
$$r_2 = \sqrt{R^2 + (z + z_0 + 2d)^2}$$
. (24c)

In above equations, **x** and **x**₀ are the field and source points, respectively; $J_0(\cdot)$ is the Bessel function of zeroth order; $v = \omega^2/g$ is the deep-water wave number.

The use of the Green's second identity to the velocity potential and Green's function can lead to a Fredholm integral equation of the second kind for ϕ_D and ϕ_R . Then, the resulting boundary-integral equations for ϕ_D and ϕ_R are expressed as follows

267
$$\alpha \phi_D(\mathbf{x}_0) - \iint_{S_b} \frac{\partial G(\mathbf{x}, \mathbf{x}_0; \omega)}{\partial n} \phi_D(\mathbf{x}) ds = \iint_{S_b} G(\mathbf{x}, \mathbf{x}_0; \omega) \frac{\partial \phi_I}{\partial n} ds, \qquad (25)$$

268 and

269
$$\alpha \phi_R(\mathbf{x}_0) - \iint_{S_b} \frac{\partial G(\mathbf{x}, \mathbf{x}_0; \omega)}{\partial n} \phi_R(\mathbf{x}) ds = \iint_{S_i} G(\mathbf{x}, \mathbf{x}_0; \omega) ds, \qquad (26)$$

in which α is a measure of the normalized solid angle, and depends on the local shape 270 271 of the boundary surface. When \mathbf{x}_0 is on S_i , α is equal to unity. When \mathbf{x}_0 is on S_b , the difficulty associated with the evaluation of α is overcome by formulating a 272 complementary problem within the interior of S_b . The higher-order boundary element 273 method (Teng and Eatock Taylor [38]) is used to solve Eqs. (25) and (26). Unknowns 274 on the body surface S_b as well as the inner free surface S_i are involved in the equations. 275 276 \mathbf{x}_0 is put on the body surface S_b as well as the inner free surface S_i , and S_b and S_i are 277 discretized into a set of curved quadrilateral or triangular elements.

As thin plates are installed within the chamber, the field point and source points are close to each other at some conditions, leading to a nearly singular integration when using the boundary element method. In order to overcome this difficulty, a self-adaptive Gauss integration method is developed in this study. Following this method, when the integration is conducted within a certain element and a nearly singular integration happens, this element will be divided into a series of finer elements with a smaller size. The subdivision will not stop until the characteristic size of the sub-elements is equal to or smaller than the distance between the source and field points. As a result, the integral in the original element is transformed into that in the sub-elements with more Gauss points. Then, after using the isoperimetric techniques, the established boundary integral equations can be transformed into a system of linear algebraic equations, and solved by standard matrix techniques.



290

Fig. 3 Element subdivision based on a self-adaptive Gauss integration method: (a) Initial element,
and (b) Sub-elements after subdivision

293

4. Calculation of the volume flux, air pressure, wave power absorption, and wave exciting force

Once the solution is obtained for the velocity potential, other physical quantities of interest can immediately be determined. Following Martins-Rivas and Mei [19], the volume flux in the wave diffraction and radiation problems can be calculated according to

300
$$q_D = \iint_{S_1} \frac{\partial \hat{\phi}}{\partial z} ds; \qquad (27a)$$

301
$$q_R = \iint_{S_i} \frac{\partial \phi_R}{\partial z} ds.$$
(27b)

With the solution of q_D , the amplitudes of the total volume flux and the oscillating air pressure can be evaluated according to the following expressions

304
$$q_c = \frac{\rho g \Lambda}{\rho g \Lambda + i \omega q_R} q_D; \qquad (28a)$$

305
$$p_c = \frac{\rho g}{\rho g \Lambda + i \omega q_R} q_D.$$
(28b)

In addition, based on the kinematic conditions on the free surface, the surface elevation amplitude, which is denoted by η , is expressed as

308
$$\eta = -\frac{1}{i\omega} \frac{\partial \phi}{\partial z}\Big|_{z=0} = -\frac{1}{i\omega} \left(\frac{\partial \phi_I}{\partial z} + \frac{\partial \phi_D}{\partial z} - \frac{i\omega}{\rho g} p_c \frac{\partial \phi_R}{\partial z}\right)\Big|_{z=0}.$$
 (29)

Following Falnes and Mciver [39], a decomposition of q_R is then made

$$-\frac{i\omega}{\rho g}q_{R} = -(C_{b} - iC_{a}).$$
(30)

Analogous to the electric circuit theory, $C_b - iC_a$ was defined as the radiation admittance in Evans and Porter [8] with C_a the radiation susceptance and C_b the radiation conductance. After inserting Eqs. (17) and (30) into Eq. (5), the following relationship is obtained

315
$$\left[\left(\chi - i\omega\mu\right) - \left(-C_b + iC_a\right)\right]p_c = q_D.$$
 (31)

316 The power captured by the air turbine becomes

317
$$W_{c} = \frac{1}{2} \chi \frac{|q_{D}|^{2}}{\left(\chi + C_{b}\right)^{2} + \left(\omega\mu + C_{a}\right)^{2}}.$$
 (32)

For given wave conditions, the optimum extraction efficiency can be achieved by 318 varying turbine parameter χ , the chamber parameter μ , or both. The chamber size and 319 the monopile geometry cannot be easily adjusted. Hence, μ cannot be optimized for a 320 321 broad range of wave conditions. As did in Martins-Rivas and Mei [19], we change the power-take off system (i.e., the parameter χ) to achieve the maximum energy extraction 322 efficiency for given dimensions of an air chamber. χ can be adjusted by some ways, 323 such as varying the angular velocity of the turbine rotor or controlling the pitch angle 324 of the turbine blades. Then, we differentiate Eq. (32) only with respect to χ to obtain the 325 326 following condition for a maximum wave power absorption

327
$$\hat{\chi}_{opt} = \sqrt{C_b^2 + (\omega\mu + C_a)^2}.$$

(33)

328 The use of Eqs. (15) and (32) gives

329
$$W_{c, \max} = \frac{1}{2} \frac{\sqrt{C_b^2 + (\omega\mu + C_a)^2} |q_D|^2}{\left[\sqrt{C_b^2 + (\omega\mu + C_a)^2} + C_b\right]^2 + (\omega\mu + C_a)^2},$$
(34)

330 and

331
$$E_{c, \max} = \frac{1}{4(R_i - a)P_{in}} \frac{\sqrt{C_b^2 + (\omega\mu + C_a)^2} |q_b|^2}{\left[\sqrt{C_b^2 + (\omega\mu + C_a)^2} + C_b\right]^2 + (\omega\mu + C_a)^2}.$$
 (35)

If we neglect the effect of the air compressibility and assume that there is no phase difference between the mass flux and the oscillating air pressure (i.e., $\mu = 0$), Eqs. (34) and (35) are reduced to

335
$$W_{c, \max} = \frac{1}{4} \frac{|q_D|^2}{\sqrt{C_a^2 + C_b^2} + C_b},$$
 (36)

336 and

337
$$E_{c, \max} = \frac{1}{8(R_i - a)P_{in}} \frac{|q_D|^2}{\sqrt{C_a^2 + C_b^2} + C_b},$$
 (37)

which are essential the same as the derivation in Evans and Porter [8].

After obtaining the velocity potential, the calculation of the wave force can be achieved by an integration of the wave pressure over the body surface. Then, we can have

$$\mathbf{f} = i\omega\rho \iint_{S_b} \left(\phi_I + \phi_D - \frac{i\omega}{\rho g} p_c \phi_R \right) \mathbf{n} ds.$$
(38)

343

342

5. Convergence test and validation

Numerical analysis is then conducted to assess the hydrodynamic performance of the combined concept. In the present study, we use $\rho = 1.025 \times 10^3$ kg/m³, $\rho_a = 1.293$ kg/m³, g = 9.807 m/s², and c = 340 m/s, respectively. In addition, the case of an annular OWC integrated into a NREL 5 MW wind turbine (Jonkman et al. [40]) is concerned. For this wind turbine, the mass of the blades as well as the nacelle is 3.5×10^5 kg, the mass of the tower is 3.4746×10^5 kg, and the radius of the monopile foundation is 3 m (i.e. a = 3 m). In addition, the height of the air volume in calm water (denoted by h_0) are fixed at $h_0 =$ 352 3.0 m in the subsequent calculation. Then, the air volume in calm water is

353
$$V_0 = h_0 \pi \left(R_i^2 - a^2 \right).$$
(39)









Fig. 6 Comparison of the calculation results of C_a , C_b , χ_{opt} , and q_D based on different mesh discretizations with $R_e = 6$ m, e = 0.06 m, h = 20 m, A = 1 m, and $\beta = 0$ for (a) C_a , (b) C_b , (c) χ_{opt} , and (d) q_D .

After obtaining C_a , C_b , and q_D , the total volume flux q_c and the oscillating air 390 pressure p_c can be determined immediately based on Eq. (28). For the purpose of 391 392 validation, an alternative solution of q_c and p_c has also been developed. The detailed deviation of this solution is introduced in the Appendix. In the alternative 393 solution, q_c and p_c are related to the incident wave and the propagation modes of 394 the radiation wave in the far field. The calculation of q_c and p_c is achieved without 395 396 the solution of the diffraction problem. A comparison of q_c and p_c based on different methods is then made. In the calculations, the turbine parameter is equal to the optimal 397 one, i.e. $\chi = \chi_{out}$, and the comparison is shown in Fig. 7. In Fig. 7, 'Direct' refers to the 398 results based on the method shown in Sections 2, 3, and 4, and 'Indirect' represents 399 those based on the alternative method. In Fig. 7, an excellent agreement is achieved 400 401 between the results based on different methods, validating the present solution.



403 Fig. 7 Comparison of the calculation results of q_c and p_c based on different methods with $R_e =$ 404 6 m, e = 0.06 m, h = 20 m, A = 1 m, $\beta = 0$, and $\chi = \chi_{opt}$ for (a) q_c , and (b) p_c .

To further examine the validity of the developed model, a comparison with the 406 reported data is then conducted. Martins-Rivas and Mei [19] studied the performance 407 of an OWC installed at the tip of a long breakwater. When the incident wave travels 408 along the thin breakwater, the effect of the breakwater can be neglected, and the case is 409 410 reduced to a truncated hollow column. An OWC in a shape of a truncated hollow column has also been considered in Deng et al. [41]. We then consider a truncated 411 hollow column with an inner radius a_c , thickness $0.002a_c$, and draft $0.4a_c$. The water 412 depth h equals $2a_c$. In the numerical simulation, 340 elements are used in each quadrant 413 (195 elements on the body surface, and 145 elements on the inner water plane area). 414 415 The present results are compared with the analytical results in Martins-Rivas and Mei [19] and Deng et al. [41]. Fig. 8 shows a good agreement between the results based on 416 417 different methods.



420 Fig. 8 Comparison of the calculation results for a truncated hollow column with analytical results 421 for (a) C_a , and (b) C_b .

423 6. Numerical results and discussion

Detailed numeral studies are conducted in this section. All the results shown in this section are in full scale, and numerical results for regular and irregular sea conditions are both presented and discussed. In addition, in the subsequent calculations, the water depth is fixed at h = 20 m.

428

429 6.1 Regular sea states

The performance of the system in regular sea states is firstly examined. In the calculations, the incident wave amplitude is fixed at A = 1 m, and wave frequency varies from 0.25 rad/s to 2.55 rad/s. The variation of q_c , p_c , W_c , and E_c with respect to wave frequency is then illustrated.





436 Fig. 9 Effect of the turbine parameter χ on q_c , p_c , W_c , and E_c with $R_e = 6$ m, d = 3 m, e =437 0.06 m, h = 20 m, A = 1 m, and $\beta = 0$ for: (a) q_c , (b) p_c , (c) W_c , and (d) E_c

Effect of the turbine parameter χ on q_c , p_c , W_c , and E_c is shown in Fig. 9 with e =439 440 0.06 m, d = 3 m, and $R_e = 6$ m. χ varies from 0.75 to 2 times the optimal turbine parameter. In addition, the wave heading is fixed at $\beta = 0$. The turbine parameter γ can 441 be adjusted by some ways, such as varying the rotational speed N. An increase of the 442 rotational speed N can cause a decrease of χ (see Eq. (7a)). From Figs. 9(a) and 9(b), 443 we note that a decrease of the turbine parameter χ reinforces the damping effect imposed 444 on the inner free surface, and then leads to a decrease of the volume flux and an increase 445 of the oscillating air pressure. In Fig. 9(c), the maximum wave power absorption is 446 attained when $\chi = \chi_{opt}$ at a certain wave frequency. In Fig. 9(d), the wave energy 447 extraction efficiency firstly continues to increase with the increase of ω until it reaches 448 a prominent peak at $\omega = 1.38$ rad/s. When ω exceeds 1.38 rad/s, it decays quickly until 449 vanished. A deviation of the turbine parameter from $\chi = \chi_{opt}$ causes a decrease of the 450 peak value and narrows down the frequency bandwidth in which high energy extraction 451 efficiency can be achieved. The effect of the air compressibility on W_c and E_c is then 452 examined. As shown in Fig. 10, the ignorance of the air compressibility (i.e. $\mu = 0$) leads 453 to an obvious overestimation of the W_c and E_c especially when the frequency is less than 454 the resonance frequency of the piston-mode motion. 455



457 Fig. 10 Effect of the air compressibility on W_c , and E_c with $R_e = 6$ m, d = 3 m, e = 0.06 m, h =458 $20 \text{ m}, A = 1 \text{ m}, \beta = 0$, and $\chi = \chi_{opt}$ for: (a) W_c , and (b) E_c .

The distribution of the wave elevation in the vicinity of the combined system at $\omega =$ 460 1.38 rad/s is calculated and shown in Fig. 11 with different turbine parameters. $\chi = +\infty$ 461 corresponds to a condition that the inner free surface open to the air, and the air turbine 462 is removed (N = 0). Around $\omega = 1.38$ rad/s, the total volume flux maximises (see Fig. 463 9(b)), and the captured wave power attains a local maximum (see Fig. 9(c)). In addition, 464 at this frequency, the radiation susceptance vanishes (see Fig. 6(a)), and the radiation 465 conductance maximises (see Fig. 6(b)). It suggests that this frequency corresponds to 466 the piston (or pumping) natural frequency of the water column in a moonpool. As a 467 result, when the inner free surface is open to the air, significantly amplified wave 468 elevation is observed at this wave frequency (see Fig. 11(a)). In Fig. 11(a), the 469 distribution of the wave elevation within the whole chamber is almost uniform, and the 470 fluid within the chamber moves like a rigid body. With a decrease of the turbine 471 parameter, more damping effect is applied on the inner free surface, and the wave 472 elevation gradually loses its uniform distribution. Its magnitude varies more and more 473 apparently along the circumferential direction, causing a breakdown of the piston-like 474 motion. 475



478 Fig. 11 Distribution of the surface elevation amplitude in the vicinity of the combined system at ω 479 = 1.38 rad/s with $R_e = 6$ m, d = 3 m, e = 0.06 m, h = 20 m, A = 1 m, and $\beta = 0$ for (a) $\chi = +\infty$, (b) χ 480 = $2\chi_{opt}$, (c) $\chi = \chi_{opt}$, and (d) $\chi = 0.5\chi_{opt}$

In the present study, vertical plates have been used to divide the whole chamber into 482 483 four sub-chambers. Based on the assumption of thin plates, an approximate solution to q_c and p_c is developed in the Appendix, in which only the zeroth-order component 484 of the incident wave is retained in the calculation. The results of q_c and p_c based 485 486 on the approximate solution have been added into Fig. 7. As shown in Fig. 7, a good agreement is achieved between the approximate solution and the complete solution. It 487 indicates that when the plates are thin, Eq. (A12) provides an effective way to evaluate 488 q_c and p_c . Eq. (A12) shows that q_c and p_c are mainly contributed by the zeroth-489 order component of the incident wave, and the wave heading has almost no effect on 490 q_c and p_c . As W_c and E_c can be evaluated directly from p_c and q_c , it suggests that 491

492 the wave heading also has a negligible effect on W_c and E_c .

We then examine the effect of the wave heading on the wave elevation in the vicinity 493 of the system. The distribution of the wave elevation amplitude at $\omega = 1.38$ rad/s is 494 shown in Fig. 12 with β varying from 0 to $\pi/4$. Under different wave headings, the wave 495 elevation within each sub-chamber distributes almost uniformly. As β increases, the 496 wave elevation in the upstream and downstream sub-chambers gets enhanced, while in 497 the remaining two sub-chambers, the wave elevation gets gradually less obvious. When 498 β increases to $\pi/4$, the distribution of the wave elevation is symmetry with respect to θ 499 $=\pi/4$, and significantly amplified wave elevation can be observed in the downstream 500 501 sub-chamber.



The effect of the plate thickness on q_c , p_c , W_c , and E_c is presented in Fig. 13 with 508 $\chi = \chi_{opt}$. The thickness of the plates is varied as e = 0.03 m, 0.06 m, 0.09 m, and 0.12 509 m, respectively. From the indirect approach in the Appendix, we note that for a given 510 incident wave, the wave power absorption by the OWC can be determined using the 511 information of the radiation wave in the far field. In the pressure-dependent radiation 512 problem, constant oscillating air pressure is applied on the inner free surface, and the 513 fluid within the chamber moves almost uniformly like a rigid body. When the plates are 514 thin, the effect of the plates on the piston-like fluid motion within the chamber as well 515 as the radiation wave in the exterior region is negligible. Therefore, the presence of the 516 thin plates makes a negligible impact on the performance of the OWC as shown in Fig. 517 13. 518



521 Fig. 13 Effect of the plate thickness on q_c , p_c , W_c , and E_c with $R_e = 6$ m, d = 3 m, h = 20 m, 522 A = 1 m, $\beta = 0$, and $\chi = \chi_{opt}$ for (a) q_c , (b) p_c , (c) W_c , and (d) E_c



0 0.2

0.6

1.0

1.4

ω(rad/s)

1.8

2.2

2.6

-

2.2

2.6

0 0.2

0.6

1.0

1.4

ω(rad/s)

1.8



Fig. 15 Effect of the chamber draft on q_c , p_c , W_c , and E_c with $R_e = 6$ m, e = 0.06 m, h = 20m, A = 1 m, $\beta = 0$ and $\chi = \chi_{opt}$ for (a) q_c , (b) p_c , (c) W_c , and (d) E_c

The effect of the exterior radius and draft of the air chamber on q_c , p_c , W_c , and E_c 532 is presented in Figs. 14 and 15 with $\chi = \chi_{opt}$. The variation of the volume flux is 533 characterized by a prominent peak (see Figs. 14(a) and 15(a)), at which significant 534 piston-like fluid motion is induced in the chamber. An increase of the exterior radius or 535 draft of the chamber can increase the amount of the entrapped fluid, and then moves 536 the frequency of the apparent piston-like fluid motion to the low-frequency region. In 537 Figs. 14(b) and 15(b), the air pressure continues to decrease with the increase of ω . In 538 addition, at the frequency of the piston-like motion, obvious enhancements of W_c can 539 be found (see Figs. 14(c) and 15(c)), and the maximum power extraction efficiency is 540 attained (see Figs. 14(d) and 15(d)). In Figs. 14(d) and 15(d), a value of E_c over unity 541 means that the capture width of the OWC exceeds $2(R_i - a)$. As remarked in Nader et 542 al. [12], due to the diffraction of the incident wave around the OWC, the adequate wave 543 power available to the device can be higher than that of the free incident wave passing 544 the device. As a result, the value of E_c can be over unity. It is also noted that a decrease 545 of the chamber width or an increase of the chamber depth can make the fluid entrapped 546 inside the chamber perform more like a rigid body, enhancing the piston-like fluid 547 motion. Then, reinforcement in the peak value of the energy extraction efficiency can 548 549 be observed with a decrease of the chamber width or an increase of the chamber depth.

In addition, in real conditions, in order to maximise the device's efficiency, it is required to keep the wave power absorption by the device as high as possible in a broader band of wave conditions. Therefore, as shown in Fig. 15(c), it is preferable to have an OWC with a shallow draft which operates better at a larger wave frequency band, than a device with an increased draft.



Fig. 16 Effect of the attached OWC on the wave force on the monopile: (a) $R_e = 6$ m, and (b) d = 3m

558

555

The effect of the attached OWC on the wave force on the monopile is then examined. The wave forces on the systems with and without an attached OWC are compared. As shown in Fig. 16, for a wide range of wave frequencies, the presence of the OWC imposes additional wave force on the system. However, at specific wave conditions, the wave forces on the OWC and the monopile can balance each other, leading to a nearly zero net force on the whole system.

565

566 6.2 Irregular sea states

The hydrodynamic performance of the combined system in irregular sea states is then examined. The real sea conditions are always complex with irregular waves. Therefore, the Jonswap spectrum has been adopted in this study to describe the characteristics of the irregular waves. Following Goda [42], the spectra density function is expressed as

571
$$S(f) = \hat{\beta}_{j} H_{s}^{2} T_{p}^{-4} f^{-5} \exp\left[-\frac{5}{4} (T_{p} f)^{-4}\right] \gamma^{\exp\left[-(T_{p} f - 1)^{2}/(2\alpha^{2})\right]},$$
(41)

572 in which

573
$$\hat{\beta}_{j} = \frac{0.06238}{0.230 + 0.0336\gamma - 0.185(1.9 + \gamma)^{-1}} (1.094 - 0.01915 \ln \gamma); \quad (42a)$$

574
$$\alpha = \begin{cases} 0.07, & f \le f_p, \\ 0.09, & f > f_p. \end{cases}$$
(42b)

In Eqs. (41), (42a) and (42b), T_p is the peak wave period; H_s is the significant wave height; $f = \omega/(2\pi)$; $f_p = 1/T_p$; γ is the peak enhancement factor, which takes a value of 3.3 in this study. The significant wave period T_s can be determined according to

578
$$T_{s} = \left[1 - 0.132(0.2 + \gamma)^{-0.559}\right]T_{p}.$$
 (43)

In the following calculations, the exterior radius and draft of the chamber are fixed at $R_e = 6$ m and d = 3 m, respectively. The thickness of the stiffening plates and that of the exterior shell of the chamber are fixed at 0.02 times the radius of the monopile, i.e. e = 0.02a = 0.06 m. In addition, in total, 7 test cases are used for irregular sea states. The test matrix for irregular sea states is listed in Table 1.

584 585

Table 1. Test matrix for irregular sea states

Case No.	$H_{s}\left(\mathbf{m} ight)$	$T_p(\mathbf{s})$	$T_{s}\left(\mathbf{s} ight)$	χ (m ³ /s/Pa)	μ (m ³ /Pa)	Spectrum
1	1.5	10	9.34	8.64×10 ⁻³	1.66×10 ⁻³	Jonswap
2	2	10	9.34	8.64×10 ⁻³	1.66×10-3	Jonswap
3	2.5	10	9.34	8.64×10 ⁻³	1.66×10 ⁻³	Jonswap
4	3	10	9.34	8.64×10 ⁻³	1.66×10-3	Jonswap
5	2	8	7.48	1.28×10 ⁻²	1.66×10-3	Jonswap
6	2	11	10.28	7.46×10 ⁻³	1.66×10-3	Jonswap
7	2	12	11.21	6.62×10 ⁻³	1.66×10-3	Jonswap

586

In the numerical implementation, a discretization of the wave spectrum is made. The wave spectrum is divided into M parts, and the frequency bandwidth d ω in each part keeps the same and is defined by

$$\mathrm{d}\omega = \frac{\omega_H - \omega_L}{M},\tag{44}$$

in which ω_{H} and ω_{L} are high and low cut-off frequencies, respectively. Then, the incident wave elevation can be expressed as a superposition of a series of the individual regular incident wave. That is

594
$$\Xi_{I}(t) = \sum_{j=1}^{M} A_{j} \cos\left[\kappa_{j} \left(x \cos\beta + y \sin\beta\right) - \omega_{j} t + \varepsilon_{j}\right] = \operatorname{Re}\left\{\sum_{j=1}^{M} A_{j} e^{i\left[\kappa_{j} \left(x \cos\beta + y \sin\beta\right) - \omega_{j} t + \varepsilon_{j}\right]}\right\}, (45)$$

595 in which

596

$$\omega_j = \omega_L + \left(j - 1 + \tau_j\right) d\omega; \tag{46a}$$

597
$$A_j = \sqrt{2S(f_j)d\omega}.$$
 (46b)

In Eqs. (45), (46a) and (46b), A_i , ω_i , and κ_i represent the amplitude, wave 598 frequency, and wavenumber of the *j*th incident wave component, respectively; 599 $f_i = \omega_i / (2\pi); \ \varepsilon_i$ is a random phase angle uniformly distributed in the range of $[0, 2\pi];$ 600 τ_i is a random number uniformly distributed in [0,1]. τ_i is introduced to impart a 601 random component to ω_i . Then, the phase-locking of the incident wave can be 602 avoided. In this study, M = 240, $\omega_L = 0.25$ rad/s, and $\omega_H = 2.65$ rad/s are used. These 603 parameters has been proved to be sufficient to represent the wave energy distribution 604 of the target spectrum. 605

Similar to the incident wave, we can express the volume flux and the oscillating airpressure as a superposition of a serious of individual components. That is

608
$$Q_{c}(t) = \operatorname{Re}\left\{\sum_{j=1}^{M} q_{c,j} e^{-i\omega_{j}t}\right\};$$
 (47a)

609
$$P_{c}(t) = \operatorname{Re}\left\{\sum_{j=1}^{M} p_{c,j} e^{-i\omega_{j}t}\right\}.$$
 (47b)

For the simulation concerning a given irregular sea state, we keep the turbine parameter χ and the chamber parameter μ both constant values, and do not vary with time. Then, after insetting Eq. (47a) and (47b) into Eq. (3), we can have

613
$$q_{c,j} = \Lambda_j p_{c,j} = \frac{\Lambda_j e^{i\varepsilon_j} q_{D,j}}{\Lambda_j - \left(-C_{b,j} + iC_{a,j}\right)},$$
 (48)

614 with

$$\Lambda_i = \chi - i\omega_i \mu, \tag{49}$$

in which $q_{D,j}$ is the complex amplitude of the volume flux in a wave diffraction problem by a regular incident wave with its frequency and amplitude being ω_j and A_j ; $C_{a,j}$ and $C_{b,j}$ are the radiation susceptance and radiation conductance at a frequency of ω_j . The calculation of $q_{D,j}$, $C_{a,j}$ and $C_{b,j}$ can be achieved by using the numerical model developed in Sections 2, 3 and 4.

Besides the volume flux and the oscillating air pressure, the wave power absorption by the system can also be evaluated for irregular sea states. After inserting Eqs. (47a), (47b), and (48) into Eq. (13), we can obtain the following expression for the wave power absorption by the system in irregular sea states

625
$$W_{c} = \sum_{j=1}^{M} \left[\frac{1}{2} \chi \frac{|q_{D,j}|^{2}}{\left(\chi + C_{b,j}\right)^{2} + \left(\omega_{j}\mu + C_{a,j}\right)^{2}} \right].$$
(50)

To illustrate the effect of the turbine parameter on the wave power absorption, the 626 variation of W_c with respect to χ is shown in Fig. 17 for Case 2, Case 5, Case 6, and, 627 Case 7, respectively. It is found that in a specific irregular sea state, the maximum wave 628 energy absorption can be achieved when χ is around $\chi_{opt_{-}T_{t}}$, in which $\chi_{opt_{-}T_{t}}$ represents 629 the optimal turbine parameter for a regular incident wave with its period equal to T_s . 630 $\chi_{opt_{-}T_{s}}$ can be evaluated using the numerical model established in Sections 2, 3 and 4, 631 and $\chi = \chi_{opt_{-}T_s}$ is adopted for the irregular sea states. The choices of χ and μ are for 632 different irregular sea states are listed in Table 1. 633





Fig. 17 Variation of the wave power absorption with respect to the turbine parameter for: (a) Case
(b) Case 5, (c) Case 6, and (b) Case 7

Figs. 18 shows the time history of the volume flux for Case 2, and Case 4 as an example. As P_c follows a similar trend as Q_c , and hence is not shown for briefly. From the time history, statistical information, such as the standard deviation, can be determined. Hereinafter, we use σ_Q and σ_P to denote the standard deviation of the volume flux and the oscillating air pressure, respectively.





Fig. 20 Effect of the peak wave period on (a) σ_o , (b) σ_p , and (c) W_c

The effect of the significant wave height is then examined. Fig. 19 shows a 652 comparison of σ_{Q} , σ_{P} and W_{c} between Case 1, Case 2, Case 3, and Case 4 in the 653 test matrix. Ren et al. [43] experimentally studied the performance of an oscillating 654 buoy type wave energy converter integrated into a NREL 5 MW wind turbine. They 655 656 suggested that to ensure the attached wave energy converter can function properly, H_s should be less than 6 m, and $H_s = 6$ m was used as a threshold for the operational sea 657 states in their work. Following Ren et al. [43], a series of significant wave heights, 658 which are all less than 6 m, i.e. $H_s = 1.5$ m, 2 m, 2.5 m and 3 m, have been adopted. As 659 shown in Fig. 19, σ_o and σ_p both continue to increase with the increase of H_s . When 660 H_s is 2 m, W_c can reach 251 kW. As H_s increases further, W_c grows in an almost 661 quadratic manner, and a larger amount of wave power can be absorbed. However, it 662 should also be noted that due to the losses of the air turbine as well as those of the 663 electric generator, the actual wave power absorption can be less than the predicted value. 664 It indicates that even when the size of the attached OWC is not large, i.e. $R_e = 6$ m and 665 d = 3 m, by adopting an optimal turbine parameter, the wave power captured by the 666 OWC can be an important supplement to the wind power production. 667

668 We then examine the effect of the peak wave period. Fig. 20 shows a comparison of

669 σ_{ϱ} , σ_{p} and W_{c} between Case 2, Case 5, Case 6 and Case 7 in the test matrix. It is 670 found that σ_{ϱ} decreases gradually as the peak wave period increases from 8 s to 12 s, 671 and σ_{p} exhibits a reversed trend. This is similar to the observation in regular sea states 672 (see Figs. 9(a, b). As shown in Fig. 9(c), $T_{p} = 8$ s, 10 s, 11 s and 12 s are all in the 673 frequency range of obvious wave power absorption. Therefore, with $T_{p} = 8$ s, 10 s, 11 674 s and 12 s, obvious wave power can be absorbed by the system in irregular sea states. 675

676 7. Conclusions

The wave interaction with a combined system consisting of an OWC and an offshore wind turbine supported by a monopile foundation is investigated. The attached OWC is of annular cross-section and consists of four fan-shaped sub-chambers. Numerical models based on HOBEM are developed to evaluate the performance of the combined system. A detailed numerical analysis is conducted for the case of an OWC integrated into a NREL 5 MW wind turbine. The main conclusions of this study are summarized as follows:

1) To examine the validity of the proposed model, two approaches, namely 'Direct' and 'Indirect', are developed in this study. A comparison of the volume flux and oscillating air pressure based on the two approaches is made, and a favorable agreement is achieved. In addition, the present numerical results also agree well with the published analytical results for OWCs of fundamental geometry.

2) Within the chamber, four stiffening plates are used to connect the exterior shell of 689 the OWC and the monopile foundation. When the plates are thin in thickness, they have 690 691 almost no effect on the wave power absorption. The maximum wave energy extraction efficiency is attained when the significant piston-like fluid motion is induced within 692 each sub-chamber. When the incident wave travels along a direction where a stiffening 693 plate is installed, the piston-like motions in different sub-chambers are with almost the 694 same amplitude. As the incident wave deviates gradually form this direction, the piston-695 696 like motions in the upstream and downstream sub-chambers get enhanced. In contrast, those in the remaining sub-chambers gets less apparent. 697

698 3) The air compressibility makes a negative effect on the wave power absorption of

the OWC especially when the wave frequency is less than the resonance frequency of 699 the piston-mode motion of the fluid in the chamber. When attaching an OWC into a 700 monopile foundation, the wave forces on the OWC and the monopile can balance each 701 other at specific wave conditions, leading to a nearly zero net force on the whole system. 702 4) The resonance frequency of the piston-mode motion depends on the geometry of 703 the OWC. The resonance frequency can be adjusted to match the incident waves by 704 varying the size of the OWC, such as the draft and exterior radius. An increase in the 705 706 width or draft of the chamber can move the resonance frequency to the low-frequency region. In addition, a decrease of the chamber width or an increase of the chamber depth 707 makes the fluid entrapped inside the chamber perform more like a rigid body. Then, the 708 piston-like fluid motion can be excited more evidently, and the corresponding energy 709 710 extraction efficiency is obviously enhanced.

5) Besides the geometric parameters, the turbine parameter is also an essential factor 711 affecting the wave power adsorption. An increase of the rotational speed of the air 712 turbine can reinforce the damping effect applied on the inner free surface, causing a 713 714 decrease of the volume flux and an increase of the oscillating air pressure. With an optimal turbine parameter, significant energy extraction efficiency can be achieved. A 715 deviation from the optimal parameter can lead to a decrease of the wave power 716 absorption, and narrow down the frequency bandwidth of the obvious energy extraction 717 efficiency. 718

719 6) An approximation approach to the volume flux and the oscillating air pressure is developed, in which a Fourier expansion of the velocity potential with respect to θ is 720 721 made, and only the zeroth-order component is retained in the calculation. A comparison between the approximate and complete solutions is made, and the comparison shows 722 that when the plates are thin, the volume flux and the oscillating air pressure within the 723 chamber is mainly contributed by the zeroth-order component of the incident waves. It 724 also shows that the wave power absorption by the proposed system is not restricted by 725 the wave direction. 726

727 7) The performance of the attached OWC in irregular sea conditions is assessed. In

an irregular sea condition, the maximum wave energy absorption can be achieved when 728 the turbine parameter is around the optimal one at the significant wave period. By 729 adopting an optimal turbine parameter, the captured wave power can exceed 250 kW 730 even when the size of the OWC is not large (i.e. $R_e = 6$ m, and d = 3 m) and in an 731 operational sea condition (i.e. $H_s = 2$ m, and $T_p = 10$ s). As wave height increases further, 732 a larger amount of wave power absorption can be expected. It indicates that the wave 733 power absorption by the OWC can be an important supplement to the combined system, 734 and the proposed combined system is a promising option for the further development. 735 In this study, we assume that the relative changes in the density and volume of the 736 737 air in the chamber are small, and a linear air turbine is considered and a linear wave theory is adopted. The characteristic nature of a hydrodynamic or thermodynamics 738 problem can be better described by a nonlinear analysis. The investigation of the effect 739 of the nonlinearity arising from the incident wave and the air turbine can be of great 740 importance, and this provides an interesting research topic for further study. 741

742

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746

747 Appendix: An alternative solution to the volume flux and air pressure

An alternative solution to the volume flux, air pressure, and wave power captured by the OWC device is introduced in the Appendix. Besides the method introduced in Sections 2, 3, and 4, it is also possible to derive a solution of q_D based on the following manner. As $\hat{\phi}$ and ϕ_R both satisfy the Laplace's equation, the application of Green's second identity to $\hat{\phi}$ and ϕ_R in the whole fluid domain leads to

753
$$\iint_{S_b + S_e + S_i + S_d + S_{\infty}} \left(\phi_R \frac{\partial \hat{\phi}}{\partial n} - \hat{\phi} \frac{\partial \phi_R}{\partial n} \right) ds = 0, \tag{A1}$$

in which S_{∞} is a cylindrical control surface surrounding the body, and its radius tends to infinity. By making use of the boundary conditions in Eqs. (20), (21) and (22), and owning to the fact that ϕ_D and ϕ_R both satisfy a Sommerfeld condition in the farfield region, Eq. (A1) can be rewritten in the following form

758
$$\iint_{S_i} \hat{\phi} ds = -\iint_{S_{\infty}} \left(\phi_R \frac{\partial \phi_I}{\partial r} - \phi_I \frac{\partial \phi_R}{\partial r} \right) ds.$$
(A2)

759 Then, we can have

760
$$q_D = \iint_{S_i} \frac{\partial \hat{\phi}}{\partial z} ds = -\frac{\omega^2}{g} \iint_{S_{\infty}} \left(\phi_R \frac{\partial \phi_I}{\partial r} - \phi_I \frac{\partial \phi_R}{\partial r} \right) ds.$$
(A3)

In the exterior region ($r \ge R_e$), we can expand the incident and radiation potentials into a Fourier series with respect to θ

763
$$\phi_{I} = \sum_{m=-\infty}^{+\infty} A_{m} J_{m} (\kappa_{0} r) Z_{0} (\kappa_{0} z) e^{im\theta}; \qquad (A4a)$$

764
$$\phi_{R} = \sum_{m=-\infty}^{+\infty} \left[B_{m0} H_{m}(\kappa_{0}r) Z_{0}(\kappa_{0}z) + \sum_{j=1}^{\infty} B_{mj} K_{m}(\kappa_{j}r) Z_{j}(\kappa_{j}z) \right] e^{im\theta}, \quad (A4b)$$

in which κ_j $(j \ge 1)$ is the *j*th positive real root of $-\omega^2 = g\kappa_j \tan(\kappa_j h); J_m(\cdot)$ is the Bessel function of order $m; H_m(\cdot)$ is the Hankel function of the first kind of order m; $K_m(\cdot)$ is the modified Bessel function of the second kind of order $m; Z_j(\kappa_j z)$ is an orthonormal function given at the interval [-h, 0]. The expression of $Z_j(\kappa_j z)$ is given by

770
$$Z_{j}(\kappa_{j}z) = \begin{cases} \frac{\cosh \kappa_{0}(z+h)}{\cosh \kappa_{0}h}, & j = 0, \\ \frac{\cos \kappa_{j}(z+h)}{\cos \kappa_{j}h}, & j \ge 1. \end{cases}$$
(A5)

The coefficients A_m and B_{mj} $(j \ge 0)$ in Eq. (A4b) can be determined according to the following expression

- $A_m = -\frac{iAg}{\omega}i^m e^{-im\beta},\tag{A6}$
- 774 and

775
$$B_{m0} = -\frac{i}{4} \frac{1}{N_0(\kappa_0 h)} \iint_{S_b + S_i} \left(\phi_R \frac{\partial}{\partial n} - \frac{\partial \phi_R}{\partial n} \right) \left[J_m(\kappa_0 r) Z_0(\kappa_0 z) e^{-im\theta} \right] ds;$$
(A7a)

776
$$B_{mj} = -\frac{1}{2\pi} \frac{1}{N_j (\kappa_j h)} \iint_{S_b + S_i} \left(\phi_R \frac{\partial}{\partial n} - \frac{\partial \phi_R}{\partial n} \right) \left[I_m (\kappa_j r) Z_j (\kappa_j z) e^{-im\theta} \right] ds, \quad j \ge 1, \quad (A7b)$$

in which $I_m(\cdot)$ is the modified Bessel function of the first kind of order m; $N_j(\kappa_j h)$ represents the inner products of the vertical eigenfunction $Z_j(\kappa_j z)$ in [-h, 0], and is defined by

780
$$N_{j}(\kappa_{j}h) = \int_{-d}^{0} Z_{j}^{2}(\kappa_{j}z) dz = \begin{cases} \frac{1}{\cosh^{2}\kappa_{0}h} \frac{h}{2} \left(1 + \frac{\sinh 2\kappa_{0}h}{2\kappa_{0}h}\right), & j = 0, \\ \frac{1}{\cos^{2}\kappa_{j}h} \frac{h}{2} \left(1 + \frac{\sin 2\kappa_{j}h}{2\kappa_{j}h}\right), & j \ge 1. \end{cases}$$
(A8)

In the indirect approach, the body surface and inner free surface is discretized into a set of elements. The radiation potentials ϕ_R on S_b and S_i are obtained by solving Eq. (26), and the integration in Eq. (A7) is performed numerically by means of a Gaussian quadrature formula.

In the far-field region, the contribution from the evanescent modes to ϕ_R can be neglected owning to the fact that $K_m(\kappa_j r)$ attenuates exponentially with distance. Then, after inserting Eq. (A4) into Eq. (A3) and using Wronskian relationships for Bessel functions, the following expression for q_D can be obtained

789
$$q_{D} = \frac{\omega^{2}}{g} \sum_{m=-\infty}^{+\infty} 4i A_{m} B_{-m0} (-1)^{m} N_{0} (\kappa_{0} h).$$
(A9)

With the solution of q_D , the total volume flux and the oscillating air pressure can be evaluated based on Eq. (28). Then, we can have

792
$$q_{c} = \frac{4i\rho\Lambda\omega^{2}}{\rho g\Lambda + i\omega q_{R}} \sum_{m=-\infty}^{+\infty} A_{m}B_{-m0} \left(-1\right)^{m} N_{0}\left(\kappa_{0}h\right);$$
(A10a)

793
$$p_{c} = \frac{4i\rho\omega^{2}}{\rho g\Lambda + i\omega q_{R}} \sum_{m=-\infty}^{+\infty} A_{m}B_{-m0} \left(-1\right)^{m} N_{0}\left(\kappa_{0}h\right).$$
(A10b)

In Eq. (A10), the total volume flux and the oscillating air pressure within the chamber is related to the incident wave as well as the radiation wave, while not depends on the diffraction wave. Eq. (A10) provides an alternative way for the evaluation of q_c and p_c .

Table A1. Variation of the magnitude of q_D (in m³/s) with respect to M with $R_e = 6$ m, e = 0.06

m, $d = 3$ m and $h = 20$ m								
$\omega = M =$	0.3 rad/s	0.6 rad/s	0.9 rad/s	1.2 rad/s	1.5 rad/s			
0	0.245×10 ²	0.487×10^{2}	0.752×10^{2}	0.131×10 ³	0.160×10 ³			
2	0.245×10 ²	0.487×10^{2}	0.752×10 ²	0.131×10 ³	0.160×10 ³			
5	0.245×10 ²	0.487×10^{2}	0.752×10^{2}	0.131×10 ³	0.160×10 ³			
10	0.245×10 ²	0.487×10^{2}	0.752×10 ²	0.131×10 ³	0.160×10 ³			

801

802 803

Table A2. Variation of the magnitude of q_D (in m³/s) with respect to *M* with $R_e = 6$ m, e = 0.06

m, $d = 4$ m and $h = 20$ m							
$\omega = M =$	0.3 rad/s	0.6 rad/s	0.9 rad/s	1.2 rad/s	1.5 rad/s		
0	0.245×10 ²	0.494×10^{2}	0.806×10 ²	0.207×10 ³	0.636×10 ²		
2	0.245×10 ²	0.494×10^{2}	0.806×10 ²	0.207×10 ³	0.636×10 ²		
5	0.245×10 ²	0.494×10^{2}	0.806×10 ²	0.207×10 ³	0.636×10 ²		
10	0.245×10 ²	0.494×10^{2}	0.806×10 ²	0.207×10 ³	0.636×10 ²		

804

When Eq. (A9) is adopted to calculate q_D , in total 2M + 1 Fourier modes (from 805 806 mode (-M) to mode M) are used in the numerical implementation. The variation of q_D with respect to M is shown in Table A1 with $R_e = 6$ m, e = c = 0.06 m, d = 3 m, and h = 0.06 m, d = 3 m, and h = 0.06 m, d = 3 m, and h = 0.06 m, d = 3 m, and h = 0.06 m, d = 3 m, and h = 0.06 m, d = 3 m, and h = 0.06 m, d = 3 m, and h = 0.06 m, d = 3 m, and h = 0.06 m, d = 3 m, and h = 0.06 m, d = 3 m, and h = 0.06 m, d = 3 m, and h = 0.06 m, d = 3 m, and h = 0.06 m, d = 3 m, and h = 0.06 m, d = 3 m, and h = 0.06 m, d = 3 m, and h = 0.06 m, d = 3 m, and h = 0.06 m, d = 0.0807 20 m. Analogous results to those in Table A1 but with d = 4 m is shown in Table A2. 808 From Tables A1 and A2, it can be seen that the results exhibit almost no variation with 809 respect to M. Other than the zeroth-order Fourier component, other components make 810 almost no contribution to q_D . In the pressure-dependent radiation problem, the air 811 pressure distributes nearly uniformly within the chamber, and the fluid within the 812 813 chamber moves like a rigid body. As the plates are very thin, their effect on the fluid motion within the chamber is not obvious. In the exterior region, the radiation wave has 814 almost no variation in the circumferential direction. Therefore, in Tables A1 and A2, 815 the results of q_D are dominated by the zeroth-order component. Then, q_D can be 816 efficiently calculated based on 817

818
$$q_D \approx 4A\omega B_{00} N_0(\kappa_0 h). \tag{A11}$$

Based on Eq. (A11), we can obtain the following expressions for the approximate evaluation of the q_c and p_c , which are given by

821
$$q_c \approx \frac{4A\rho g\omega\Lambda}{\rho g\Lambda + i\omega q_R} B_{00} N_0 (\kappa_0 h); \qquad (A12a)$$

822
$$p_c \approx \frac{4A\rho g\omega}{\rho g \Lambda + i\omega q_R} B_{00} N_0 (\kappa_0 h).$$
(A12b)

823

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